

Lecture 7

Closed-Loop Analysis

Part I: Laplace Domain Analysis

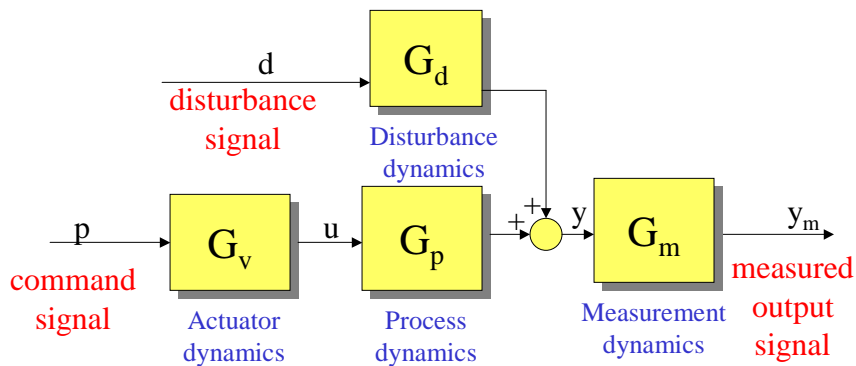
CHE4400

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Overview



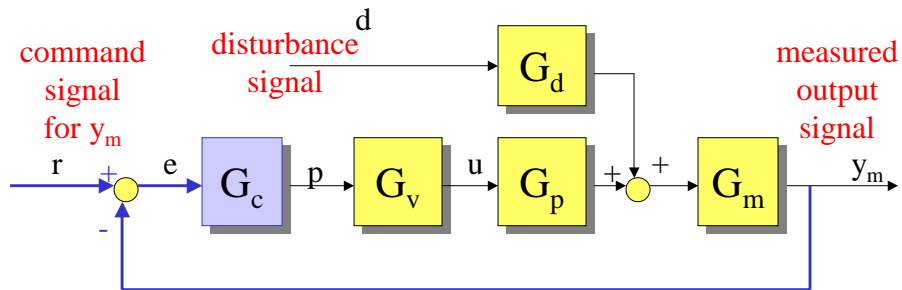
$$\frac{\hat{y}_m(s)}{\hat{p}(s)} = G_m G_p G_v$$

- The gain may not be 1
- The dynamics may be too slow, too oscillatory or unstable.

$$\frac{\hat{y}_m(s)}{\hat{d}(s)} = G_m G_d$$

- The gain may not be 0
- The dynamics may be too slow.....

Overview



$$\frac{\hat{y}_m(s)}{\hat{r}(s)} = \frac{\underbrace{G_m G_p G_v G_c}_{\mathbf{G}}}{1 + G_m G_p G_v G_c} = \frac{GG_c}{1 + GG_c}$$

• Different from open-loop!
• Depends on G_c

$$\frac{\hat{y}_m(s)}{\hat{d}(s)} = \frac{\underbrace{G_m G_d}_{\mathbf{G}_{md}}}{1 + G_m G_p G_v G_c} = \frac{G_{md}}{1 + GG_c}$$

Analysis and Design Problems

- Analysis: Given particular G , G_{md} and G_c
 - Are the closed-loop dynamics stable? ← *Locations of poles and zeros of the closed-loop transfer functions*
 - Speed of response? Damping? ← *Locations of poles and zeros of the closed-loop transfer functions*
 - Gains for y_m/r and y_m/d
- Design: Given particular G and G_{md} , choose (“design”) G_c so that
 - the closed-loop dynamics are stable.
 - y_m/r has a gain of ~ 1 and y_m/d has a gain of ~ 0 .
 - the dynamics are sufficiently fast (but not too fast) and smooth (without excessive oscillation).

Model Used for Analysis and Design

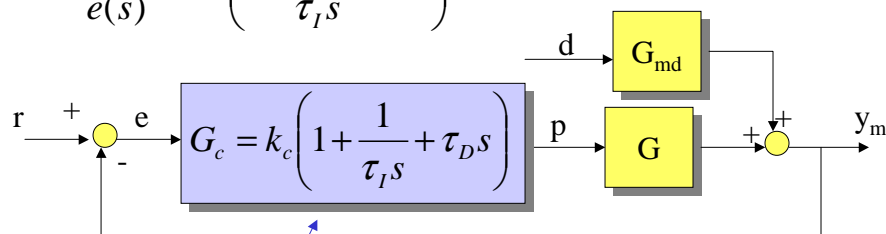
- Case I (Less Frequent)
 - From a fundamental model, perform linearization and Laplace transform of the linearized ODEs to find $G_p(s)$ and $G_d(s)$
 - Find actuator and measurement dynamics G_v and G_m to obtain $G = G_m G_p G_v$ and $G_{md} = G_m G_d$
- Case II (More Frequent)
 - The composite model G is fitted to data of y_m obtained by perturbing p (e.g., by making a step change).
 - G_{md} is often assumed to be same as G (as in IAE/ISE/ITAE tuning rules).

PID Controller

$$p(t) = \bar{p} + k_c \left(e(t) + \frac{1}{\tau_I} \int_0^t e(t') dt' + \tau_D \frac{de}{dt} \right) \Rightarrow$$

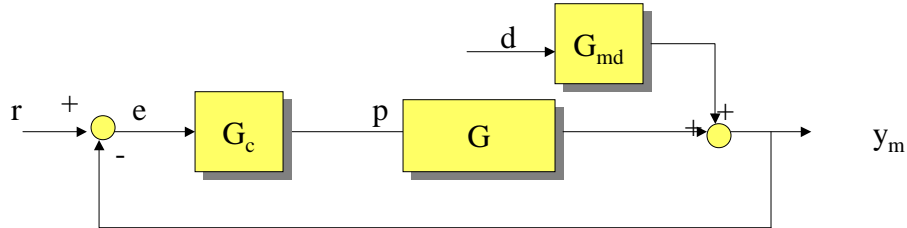
$$\hat{p}'(s) = k_c \left(\hat{e}(s) + \frac{1}{\tau_I s} \hat{e}(s) + \tau_D s \hat{e}(s) \right) \Rightarrow$$

$$\frac{\hat{p}'(s)}{\hat{e}(s)} = k_c \left(1 + \frac{1}{\tau_I s} + \tau_D s \right)$$



However, G_c can be chosen as any transfer function in general.

Calculation of Closed-Loop Functions



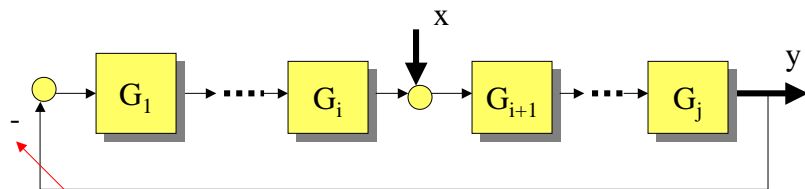
$$\hat{y}_m = G_{md}\hat{d} + G\hat{p} \text{ and } \hat{p} = G_c(\hat{r} - \hat{y}_m) \Rightarrow$$

$$\hat{y}_m = G_{md}\hat{d} + GG_c(\hat{r} - \hat{y}_m) \Rightarrow (1 + GG_c)\hat{y}_m = G_{md}\hat{d} + GG_c\hat{r}$$

$$\frac{\hat{y}_m(s)}{\hat{d}(s)} = \frac{G_{md}}{(1 + GG_c)} \text{ and } \frac{\hat{y}_m(s)}{\hat{r}(s)} = \frac{GG_c}{(1 + GG_c)}$$

See the convenience the Laplace transform affords us?

Calculation of Closed-Loop Functions - General



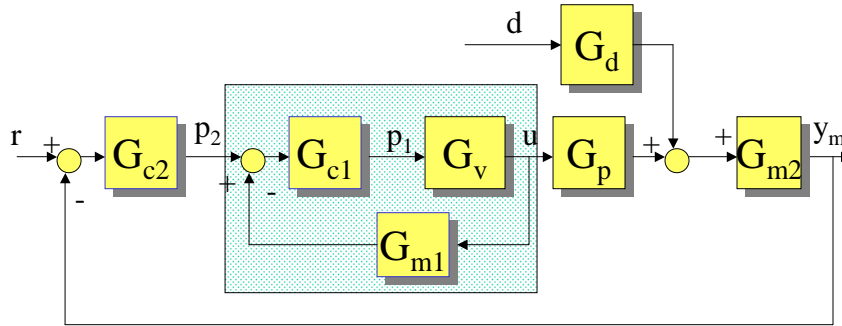
Assume "negative feedback"

open-loop path from x to y

$$\frac{\hat{y}(s)}{\hat{x}(s)} = \frac{G_{i+1}G_{i+2} \cdots G_j}{1 + G_1G_2 \cdots G_j} = \frac{G_{OL}}{1 \oplus G_{CL}}$$

Product of all blocks inside the loop

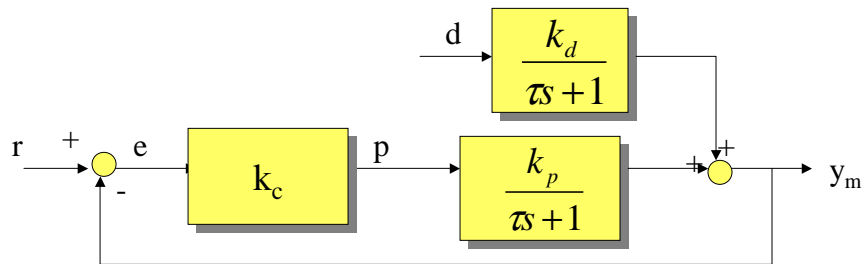
Handling of Cascaded Loops



$$\frac{\hat{u}(s)}{\hat{p}_2(s)} = \frac{G_{c1}G_v}{1 + G_{c1}G_vG_{m1}} = G_{vnew} \quad \text{New actuator dynamics modified by the inner loop!}$$

$$\frac{\hat{y}_m(s)}{\hat{r}(s)} = \frac{G_{m2}G_pG_{vnew}G_{c2}}{1 + G_{m2}G_pG_{vnew}G_{c2}}; \quad \frac{y_m(s)}{d(s)} = \frac{G_{m2}G_d}{1 + G_{m2}G_pG_{vnew}G_{c2}}$$

Analysis of P-only Control



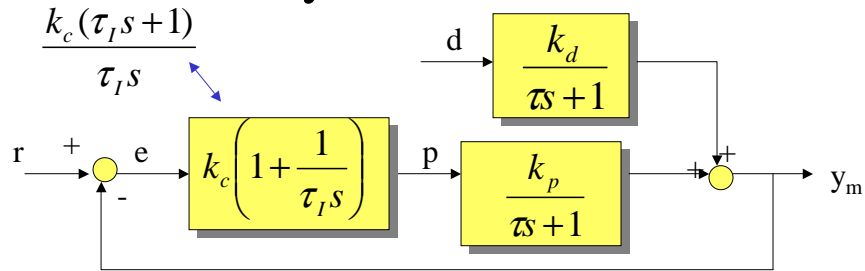
$$\frac{\hat{y}_m(s)}{\hat{r}(s)} = \frac{k_c k_p}{1 + \frac{k_c k_p}{\tau s + 1}} = \frac{k_c k_p}{\tau s + 1 + k_c k_p} = \frac{\frac{k_c k_p}{1 + k_c k_p}}{\frac{\tau}{1 + k_c k_p} s + 1}$$

- Gain is not 1 unless $k_c = \infty$
- Time constant decreases with increasing k_c

$$\frac{\hat{y}_m(s)}{\hat{d}(s)} = \frac{\frac{k_d}{\tau s + 1}}{1 + \frac{k_c k_p}{\tau s + 1}} = \frac{k_d}{\tau s + 1 + k_c k_p} = \frac{\frac{k_d}{1 + k_c k_p}}{\frac{\tau}{1 + k_c k_p} s + 1}$$

- Gain is not 0 unless $k_c = \infty$

Analysis of PI Control



$$\frac{\hat{y}_m(s)}{\hat{r}(s)} = \frac{\frac{k_c k_p (\tau_I s + 1)}{\tau_I s (\tau s + 1)}}{1 + \frac{k_c k_p (\tau_I s + 1)}{\tau_I s (\tau s + 1)}} = \frac{k_c k_p (\tau_I s + 1)}{\tau_I s (\tau s + 1) + k_c k_p (\tau_I s + 1)}$$

$$= \frac{(\tau_I s + 1)}{\frac{\tau_I \tau}{k_c k_p} s^2 + \frac{1 + k_c k_p}{k_c k_p} \tau_I s + 1}$$

- Gain = 1 always! No offset.
- 2nd order dynamics
- Underdamped dynamics for very small τ_I

Characteristic Equation

$$1 + GG_c = 0$$

- Roots of the above equation are the poles of the closed-loop functions (important information for analyzing closed-loop dynamics)
- For stability, make sure all the roots are in the Left-Half-Plane (negative real parts)
 - Can be checked by Routh's test (14.6.3 of the textbook)
 - Or by Direct Substitution (14.6.4)

Example: Routh's Test

Main Idea: Form a Routh array to see if any roots are in the RHP.

$$1 + \frac{6K_c}{(2s+1)(4s+1)(6s+1)} = 0 \Rightarrow$$

$$48s^3 + 44s^2 + 12s + (1 + 6K_c) = 0$$

$$\frac{44 \times 12 - 48(1 + 6K_c)}{44}$$

$$\frac{\left(\frac{120}{11} - \frac{72}{11}K_c\right) \times (1 + 6K_c) - 44 \times 0}{\left(\frac{120}{11} - \frac{72}{11}K_c\right)}$$

$$\begin{array}{r|rr} 48 & 12 & \\ \hline 44 & 1 + 6K_c & \\ \hline \frac{120}{11} - \frac{72}{11}K_c & 0 & \\ \hline 1 + 6K_c & & \end{array}$$

Must be all positive
for closed-loop
stability!

$$\frac{120}{11} - \frac{72}{11}K_c > 0 \Rightarrow K_c < \frac{5}{3}$$

$$1 + 6K_c > 0 \Rightarrow K_c > -\frac{1}{6}$$

Example: Direct Substitution

Main Idea: At the limits of instability, the closed-loop poles will be on the imaginary axis (between LHP and RHP)

$$48s^3 + 44s^2 + 12s + (1 + 6K_c) = 0 \xrightarrow{s=j\omega}$$

$$-48j\omega^3 - 44\omega^2 + 12j\omega + (1 + 6K_c) = 0 \rightarrow$$

$$(-48\omega^3 + 12\omega)j + [-44\omega^2 + (1 + 6K_c)] = 0 \rightarrow$$

$$-48\omega^3 + 12\omega = 0 \rightarrow \omega = 0, K_c = -1/6$$

$$-44\omega^2 + (1 + 6K_c) = 0 \rightarrow \omega = \pm 1/2, K_c = 5/3$$

This method works with a system with time delay
(see Example 14.10 in your textbook). Routh's method does not.

Root Locus Diagram

Main Idea: Shows how the closed-loop poles move with different controller gain values.

